

Introduction

- Thermal barrier coatings (TBCs) are applied to turbine blades to insulate superalloy blades in jet engines
- Yttria-stabilized zirconia (YSZ) is the standard material for TBCs
- Coating is applied through Electron-Beam Physical Vapor Deposition (EB-PVD) resulting in a columnar microstructure
- Deposits, such as sand or volcanic ash, enters the engine, melts and ingresses into the coating during operation [1,2]







Images of turbine blades have been used with permission by the German Aerospace Center (DLR)

- **a.** Depiction of a jet engine with the turbine section labelled
- **b.** Image of a turbine blade exposed to CMAS during operation
- **c.** Close-up of coating failure due to CMAS exposure

Motivation and Objective

Motivation

- During CMAS ingression, yttria is depleted
- Causes phase transformations within the coating
- Increases strain and risk of failure
- Decreases lifetime expectancy
- High-energy X-ray diffraction can quantify phases present

Objective

• Determine the effect of CMAS ingression on TBC by measuring the phase volume fraction (PVF) of the monoclinic phase (mPVF) present in the coating

2D XRD & Phase Volume Fraction Theory

- High-energy X-rays transmit through material, producing unique Debye-Scherrer rings according to the lattice spacing of the present crystalline phases
- Rings hold information regarding the amount of phases present, texture, and strain within the material
- Phases can be identified through reference XRD databases



CMAS Ingression Study on EB-PVD Thermal Barrier Coatings using Synchrotron X-Ray Diffraction

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- the coating and CMAS
- The surface of the samples react more to the CMAS than further into the coating due to CMAS deposition starting at the surface





Conclusion

• Depth of CMAS ingression can be quantified through mPVF CMAS has a significant effect on the amount of monoclinic phases present within the coating

- This effect can be seen throughout the depth of the coating
- mPVF decays exponentially throughout the depth of the coating
- Annealing time strongly effects the tetragonal to monoclinic phase transformation
 - Longer annealing times result in higher monoclinic phase
 - CMAS is able to ingress further into and react with the coating

Future Work

Temperature Effects of CMAS

• Determine the effect of annealing temperature at which the deposit ingresses the coating

Residual Strain Analysis

 How does the deposit ingression effect the strain state of the coating

In-situ CMAS Ingression Synchrotron X-ray Diffraction Study

 65YSZ TBC system samples with CMAS deposited on top

Sample is heated during XRD measurements for real-time ingression

Acknowledgements & References

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